

Constrained Geometric Parameterization by Mesh Morphing for a Catamaran Foils Optimization Procedure

Ubaldo Cella^{1,2}, Marco Evangelos Biancolini^{2,3},
Alberto Clarich⁴, Francesco Franchini⁵

¹ Design Methods *aerospace consulting*

² University of Rome "Tor Vergata"

³ RBF Morph

⁴ ESTECO

⁵ EnginSoft

Design by optimization

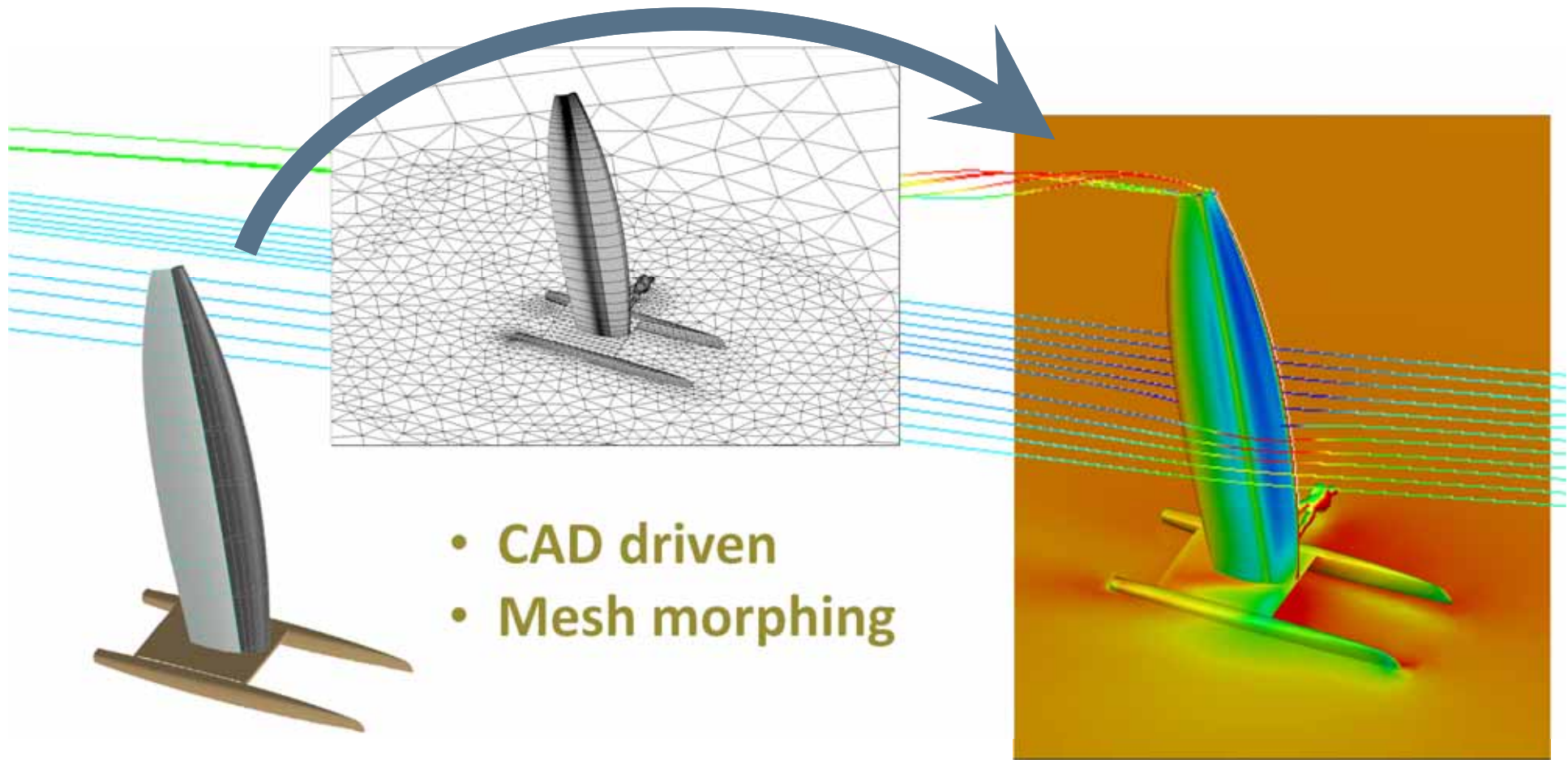
Optimization environment

Geometric
parameterization



Numerical analysis

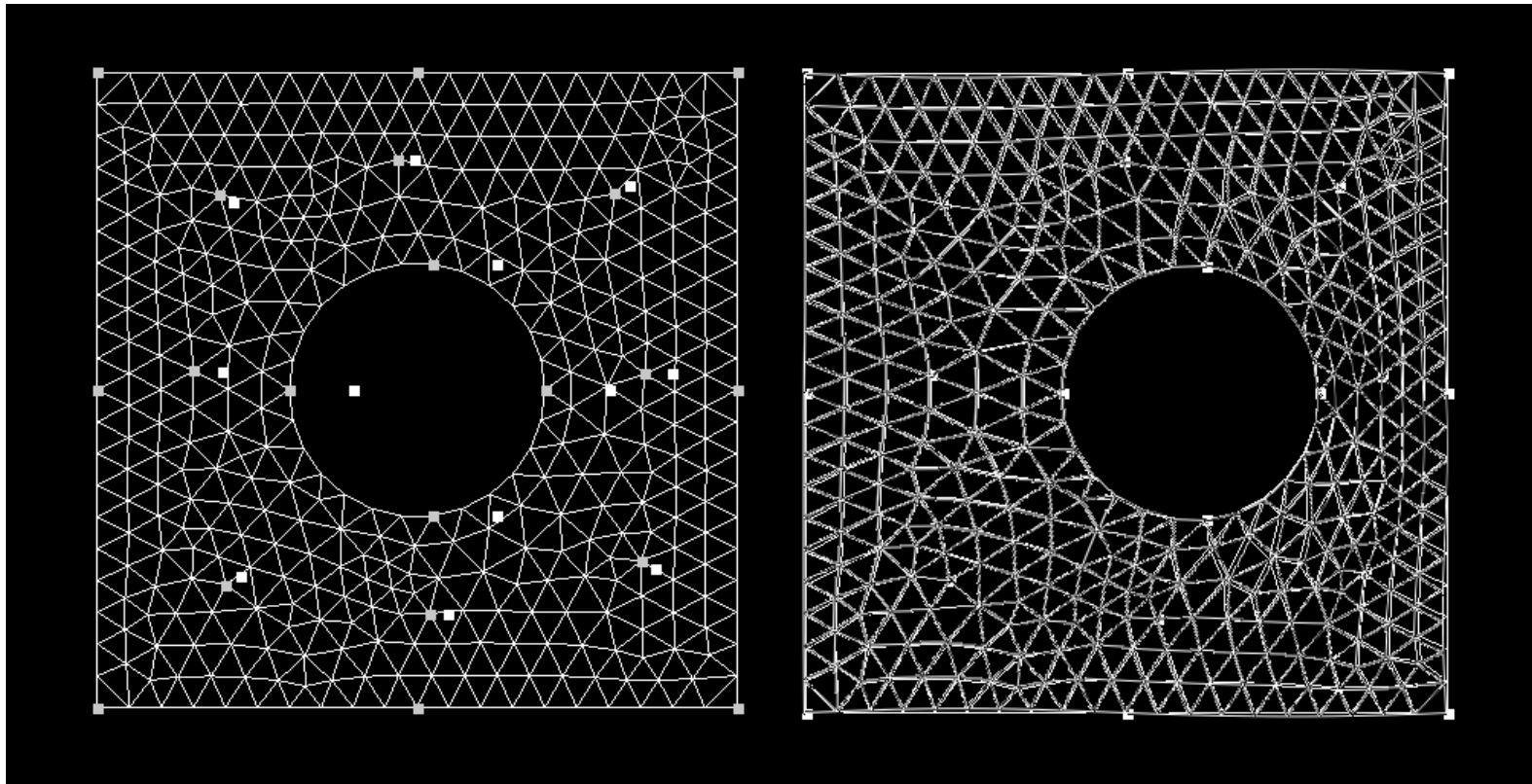
Domain adaptation



CAD to mesh

- Main advantages
 - Accurate geometry quality control
 - High constraints setup flexibility
 - No “back to CAD” required
- Main disadvantage
 - Complex and not generalizable setup
 - Highly skilled CAD user required
 - Robustness
 - Remesh required
 - Structured grids
 - Simple geometries

Mesh morphing



RBF for mesh morphing

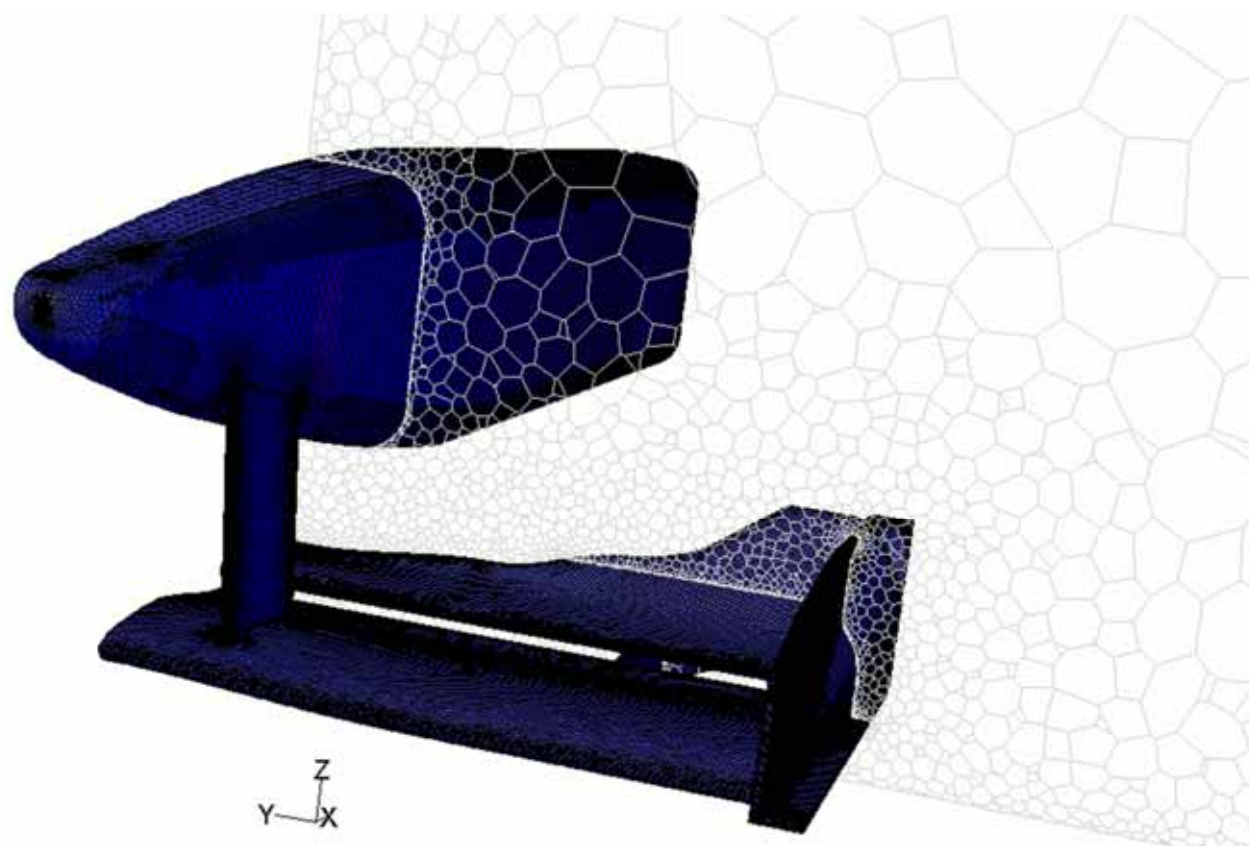
- **Radial Basis Functions** (RBF) can be used to drive mesh morphing (smoothing) from a list of source points and their displacements.
 - Surface shape changes (exact nodes control)
 - Volume mesh smoothing.
- RBF are recognized to be one of the **best mathematical tool** for mesh morphing.

RBF mesh morphing

- Main advantages
 - No re-meshing
 - Can handle any kind of mesh
 - Can be integrated in the CFD solver
 - Highly parallelizable
 - Robust process
- Main disadvantage
 - Computationally expensive (HPC for large grids)
 - Back to CAD procedure required
 - Uncertainness in setting up complex constrained geometric problems

(rbf-morph)TM

Welcome to the World of Fast Morphing!



RBF Morph software line

- add-on for ANSYS **Fluent** CFD solver
- Stand alone (GUI+TUI)
 - OpenFOAM, Nastran, elsA, CFD++, StarCCM+, CGNS, NASTRAN
- ANSYS Mechanical ACT module
- HPC RBF general purposes library
 - It is the kernel of RBF Morph (parallel, GPU)



DESIGN
METHODS™

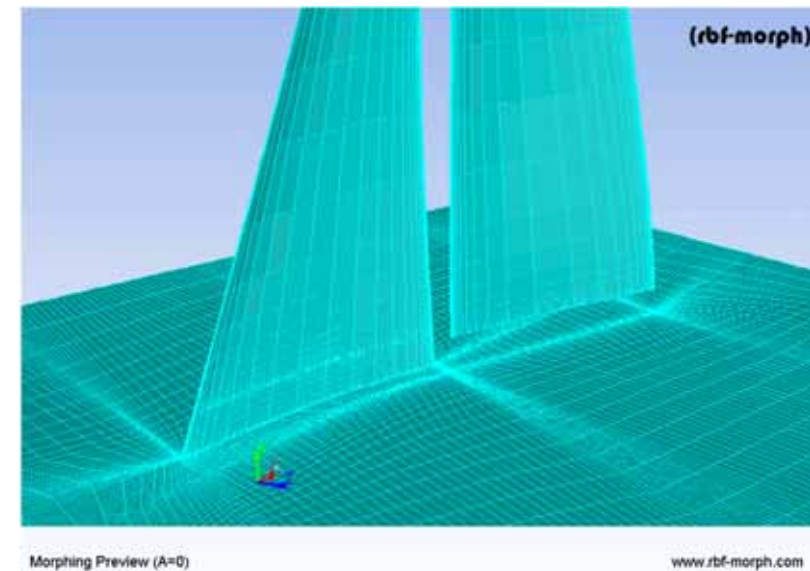
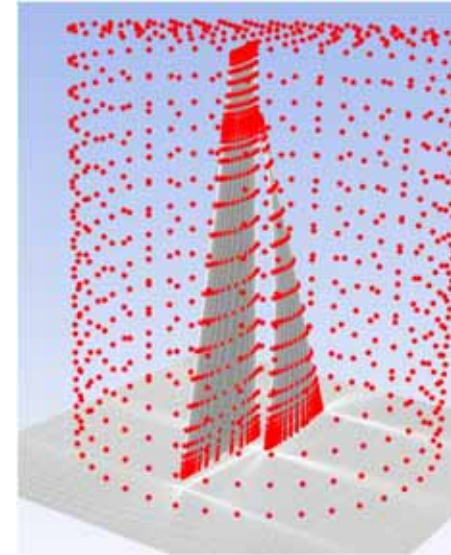
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2017, 6 - 7 November

How it works

- Setup
 - Select fixed and moving walls by source points
 - Prescribe the displacements (or a combination of)
- Fitting
 - Solution and storing of the RBF system
- Smoothing
 - Application of the computed morphing actions on surfaces and volume



Solver performance samples

- **14 mill.** cells, 60.000 points, PC **4 cpu** 2.67 GHz
 - fitting time: **53 sec.** (serial)
 - smoothing: **3.5 min.**
- **50 mill.** cells, 30.000 points, HPC **140 cpu**
 - fitting time: **25 sec.** (serial)
 - smoothing: **1.5 min.**
- **100 mill.** cells, 200.000 points, HPC **256 cpu**
 - fitting time: **25 min.**
 - smoothing: **5 min.**
- Largest fitted cloud **2 mill.** points on **32 cpu** in **3 hours.**
- Largest model morphed (in our knowledge) **700 mill.** cells on **768 cpu** in **45 min.**



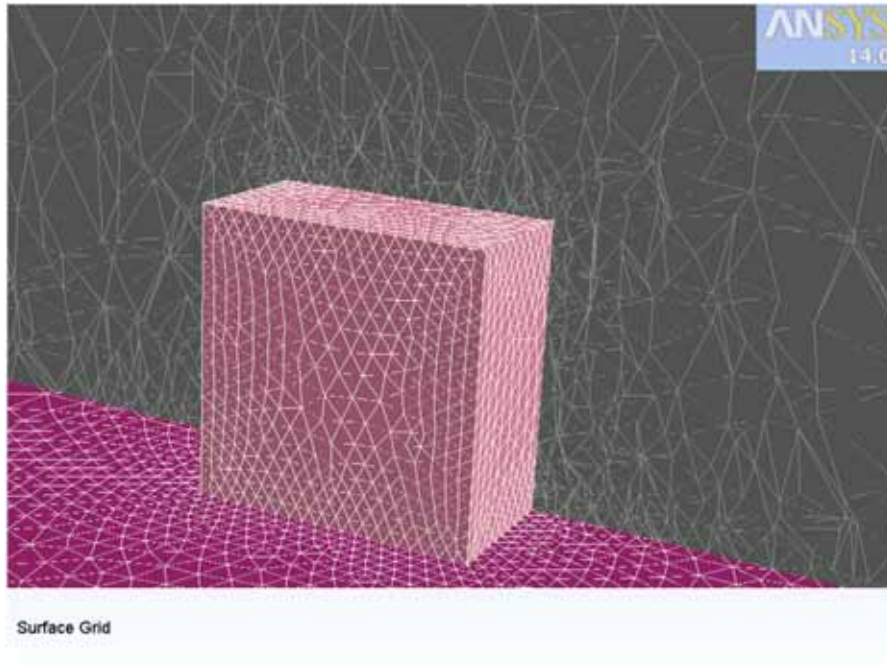
DESIGN
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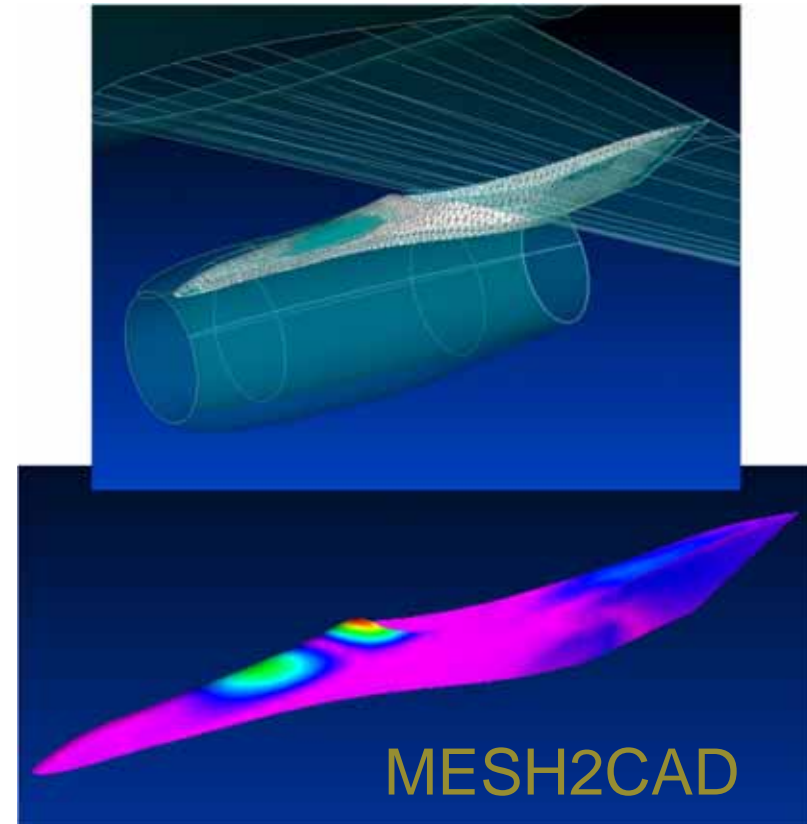


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CAD input and output



STL target



Objectives of this work

- Test the capability of mesh morphing approach to manage **complex constrained shape parameterization**
- Verify its efficiency when coupled with leading technologies in an **optimization environment**
- Develop a **challenging test pilot problem** to demonstrate the capability of the proposed approach

Partners



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Aerospace Engineering - www.designmethods.aero

A-Class cat foils design

The fastest single handed racing boats

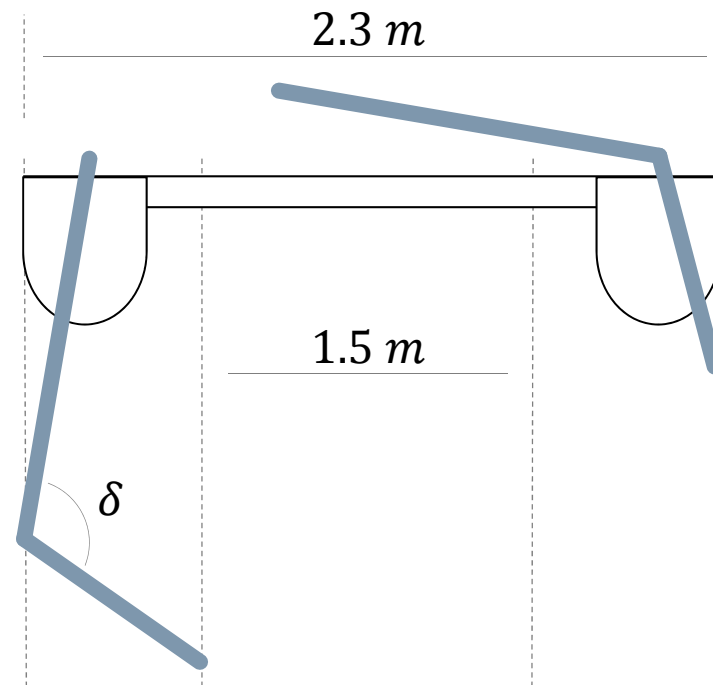


Geometric constraints

A-Class Rules

8.1 - No part of each hull or hull appendages below the waterline shall be less than 0.75 meters from the centre line

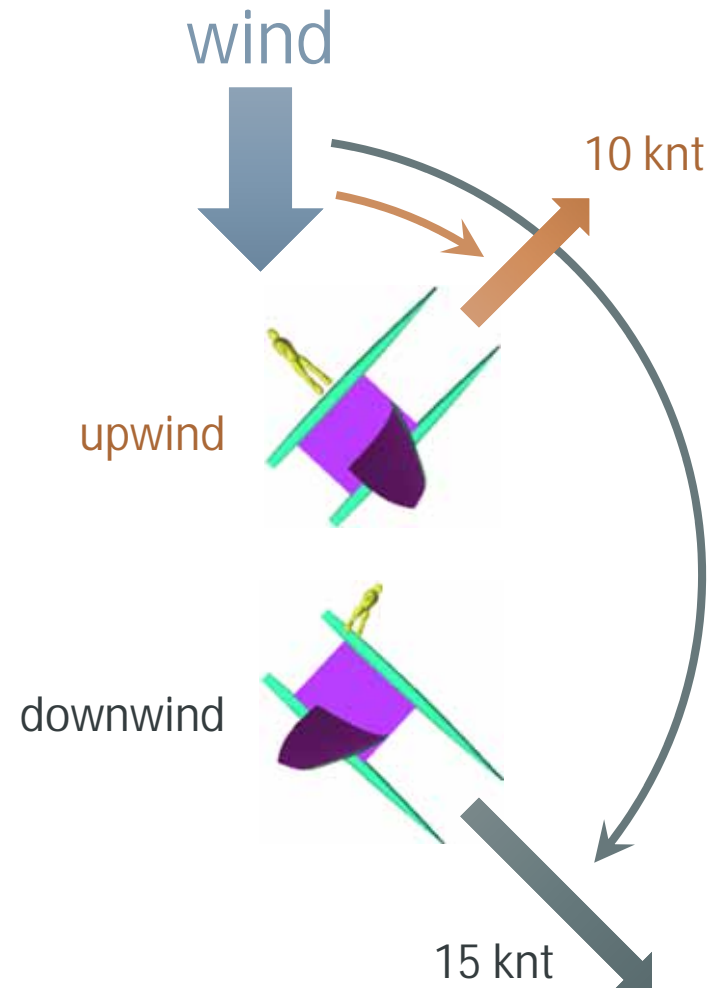
8.2 - Movable and retractable hull appendages shall be inserted from the top or be capable of being fully retractable into the hull.



Design conditions

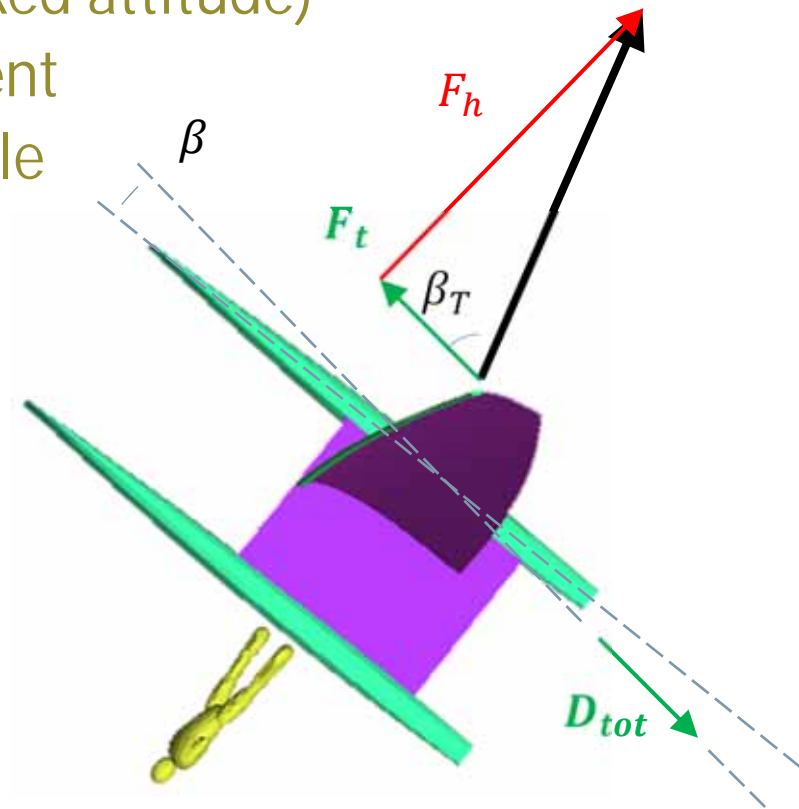
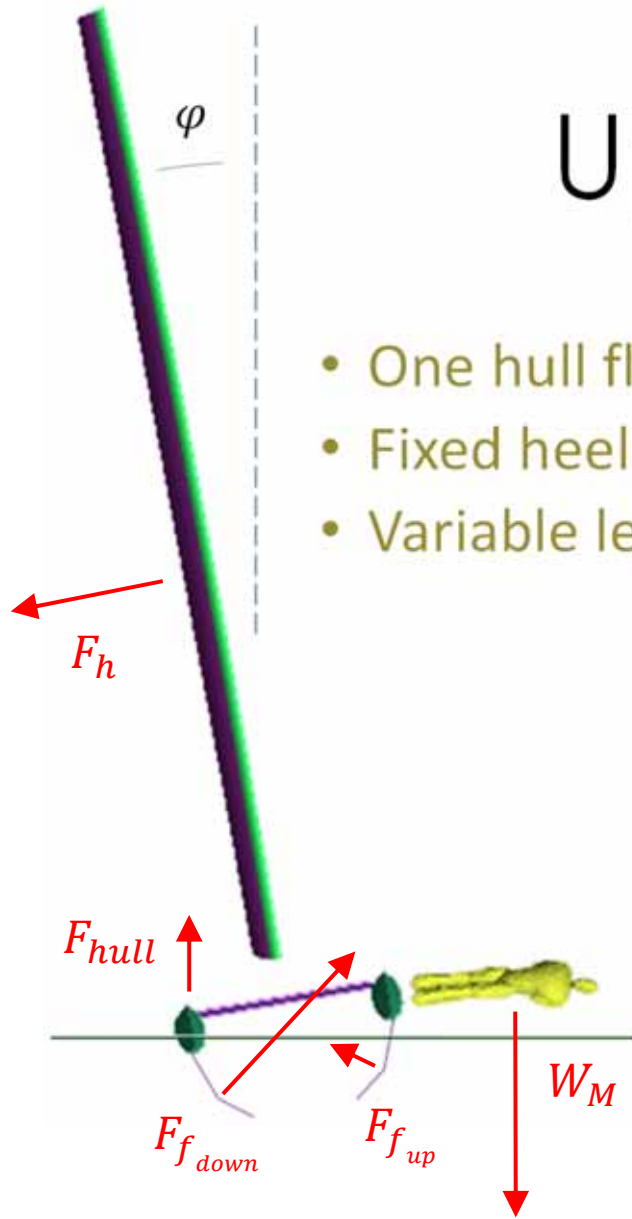
Total displacement = 170 Kg
heeling angle = 5 deg

- Upwind sailing
 - “traditional” sailing
 - Boat speed = 10 knots
 - fixed sinkage
 - free leeway angle
- Downwind sailing
 - “foiling” sailing
 - Boat speed = 15 knots
 - leeway angle = 3 deg
 - free sinkage

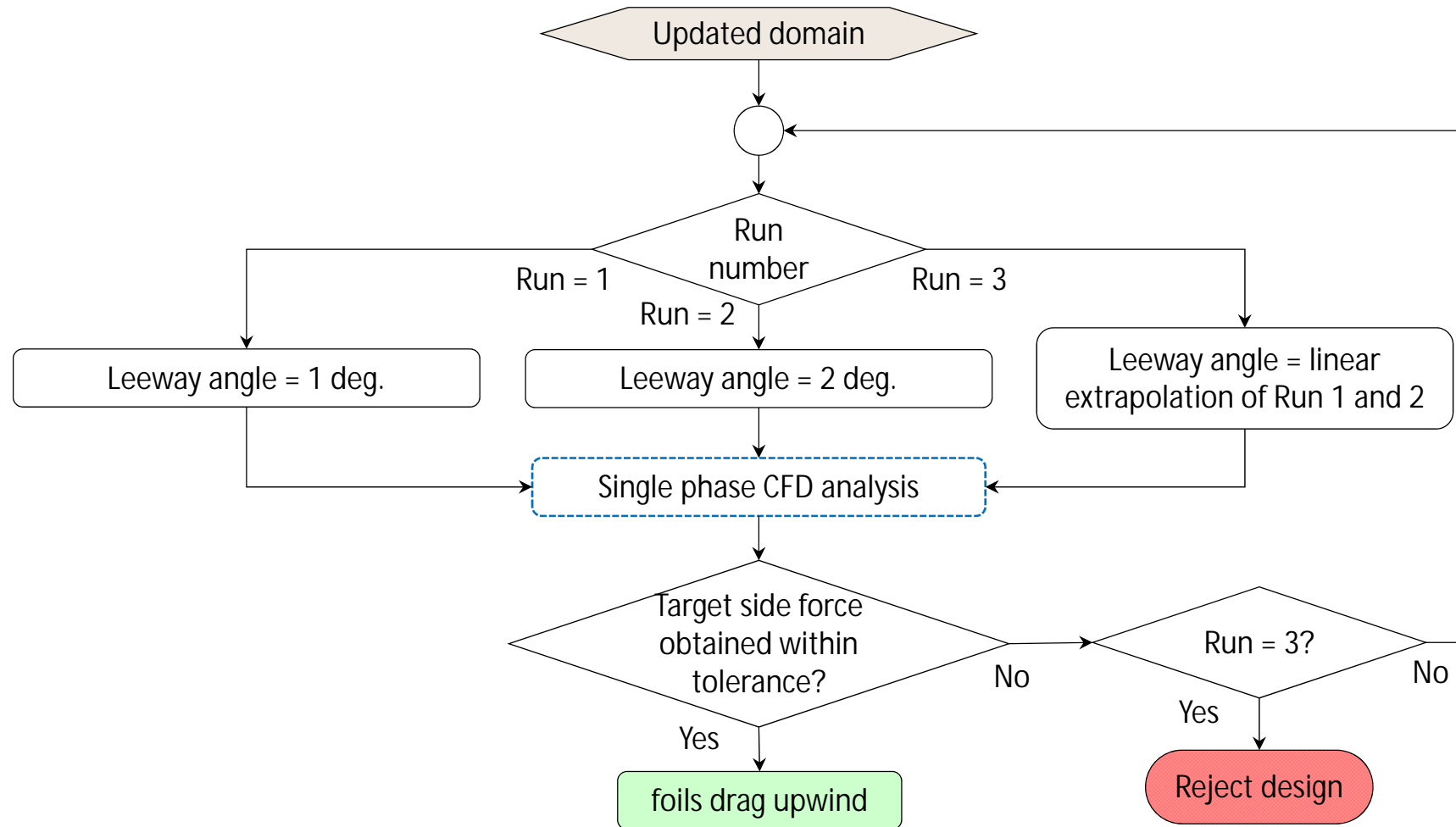


Upwind equilibrium

- One hull floating (fixed attitude)
- Fixed heeling moment
- Variable leeway angle

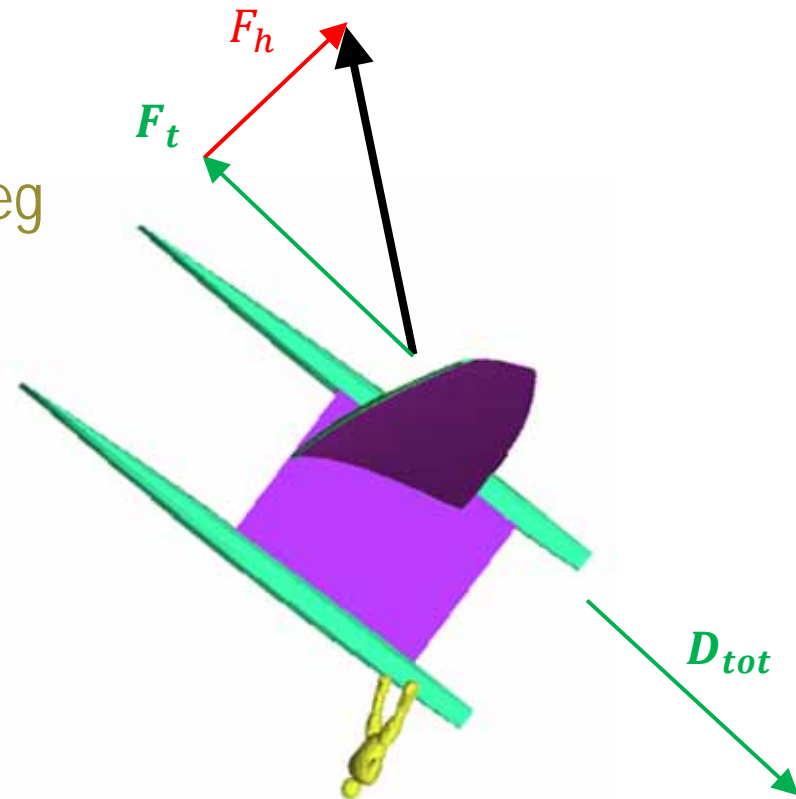
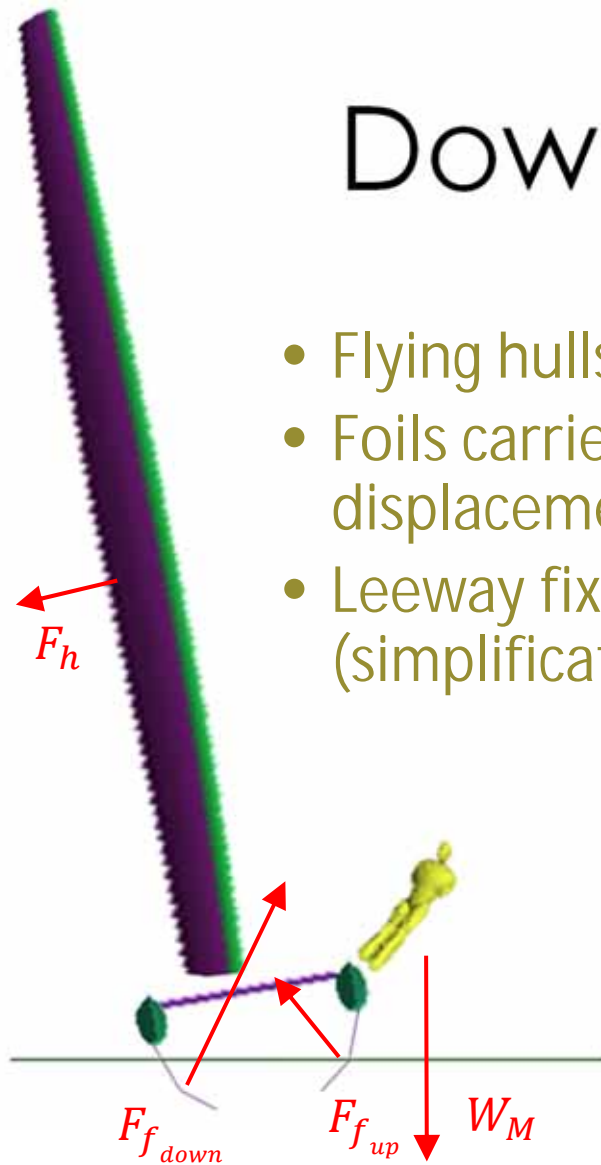


Upwind analysis

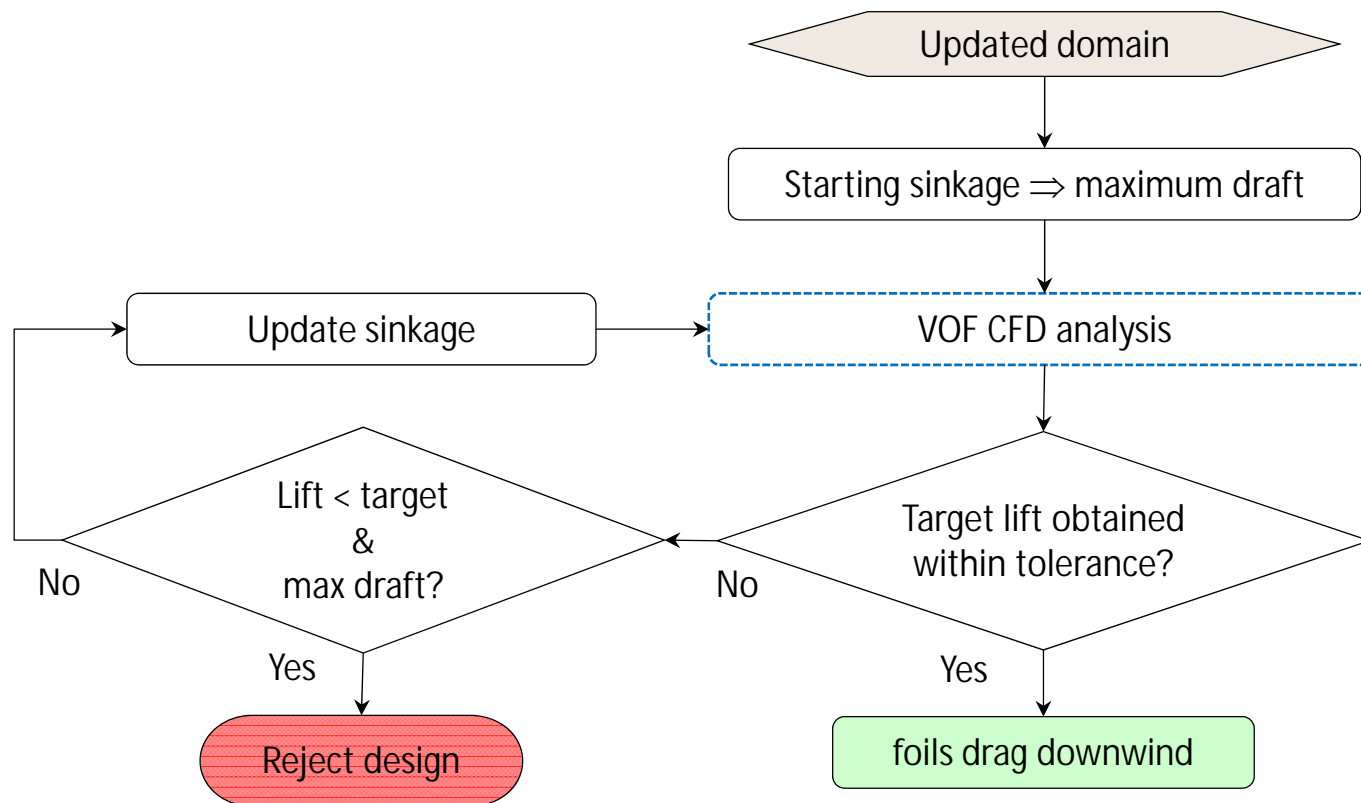


Downwind equilibrium

- Flying hulls (foiling)
- Foils carries 70% of displacement
- Leeway fixed to 3 deg (simplification)

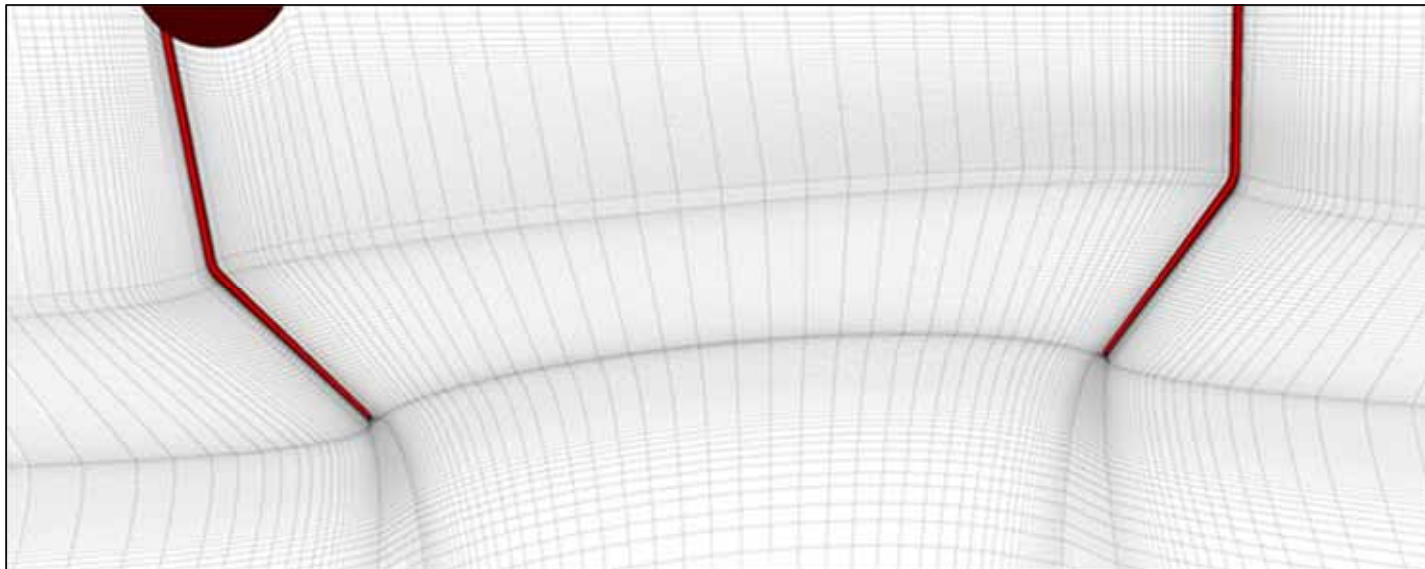
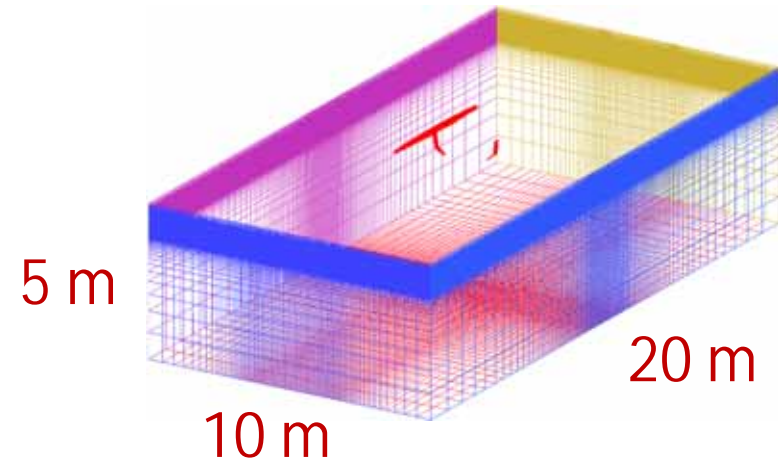


Downwind analysis



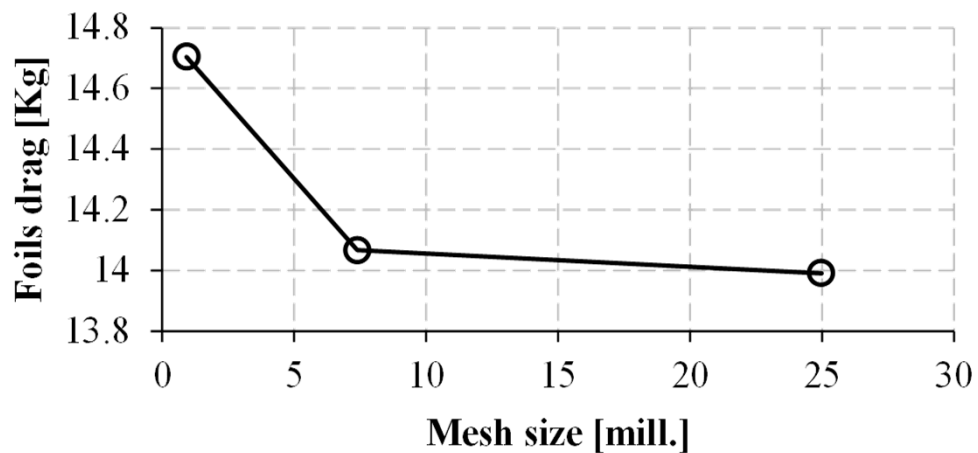
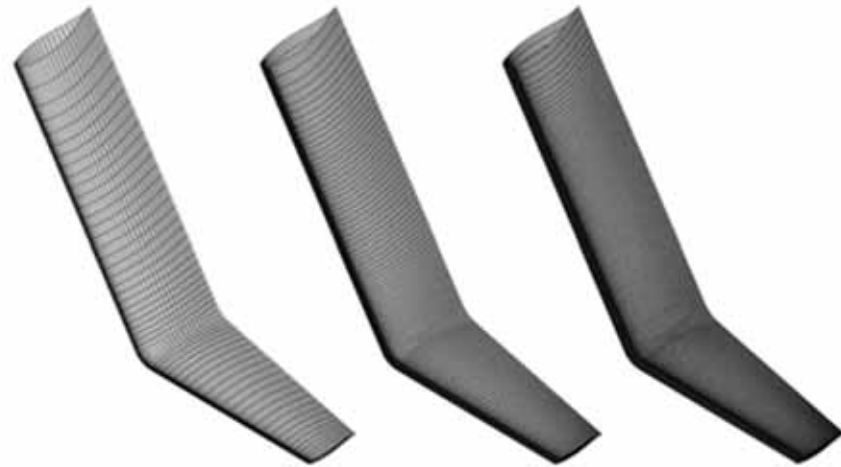
Computational domain

- Structured hexa
 - Inviscid hull
 - Wall functions on foils (fully turbulent BL)



Grid sensitivity analysis

- Level 1 = **1** millions
- Level 2 = **7.5** millions
- Level 3 = **25** millions



} order of 5 %
(downwind analysis)

Front shape parameters

- Front shape variables of design
 1. total foil draft
 2. outer segment cant angle
 3. inner segment angle respect to vertical

```
(define outang (- inpoutercant (+ inpinnercant 90))) ; Horizontal angle of the outer foil section [deg]
(define sweep (- insweep startswEEP))
(define outercant (- inpoutercant startoutercant))
(define innercant (- inpinnercant startinnercant))
:(define tip (- (* outertaperatio startrootchord) starttipchord)) ; [cm] (vecchio setup)
(cond
  ((= outertaperatio 0.4) (define tip -1))
  ((= outertaperatio 0.8) (define tip :))
  (else (define tip 3))
)
(define innerspan (/ (- (* maxspan (tan (/ (* pi outang) 180))) tipdraft) (- (* (sin (/ (* pi inpinnercant) 180)) (tan (/ (* pi outang) 180))) (cos (/ (* pi inpinnercant) 180)))))
(define outerspan (/ (- maxspan (* innerspan (sin (/ (* pi inpinnercant) 180))) (cos (/ (* pi outang) 180))))
(define innerspanamp (- startinnerspan innerspan) ; subtracting inner foil span [cm]
(define outerspanamp (- startouterspan outerspan) ; subtracting outer foil span [cm]
```

Scheme script

$$\Delta L_{out}$$

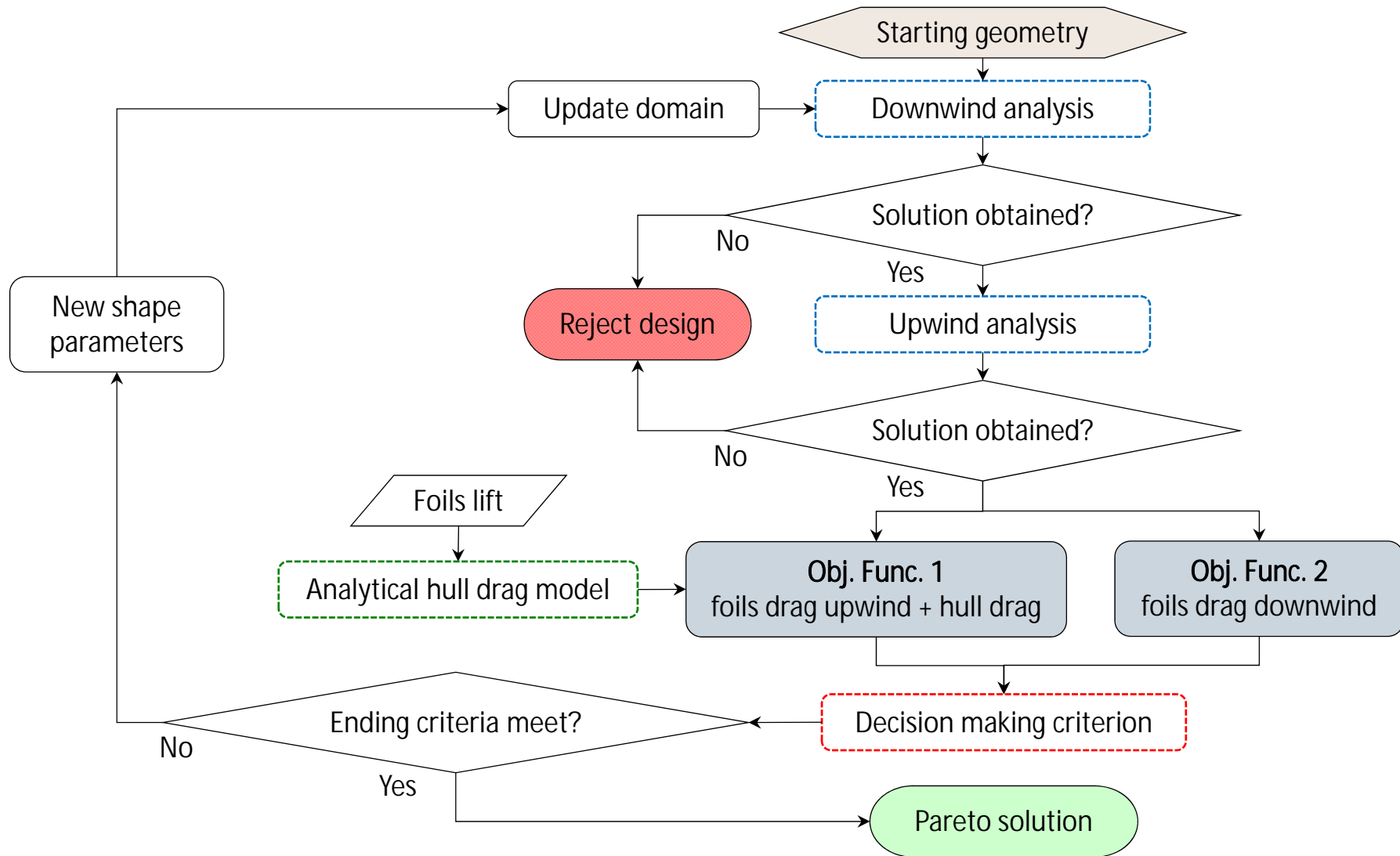
$$\Delta \delta_{out}$$

$$\Delta \delta_{in}$$



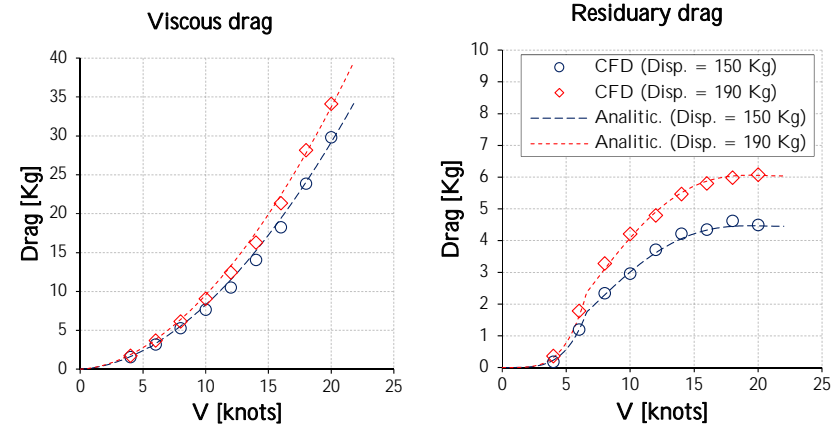
Planform parameterization

Optimization workflow

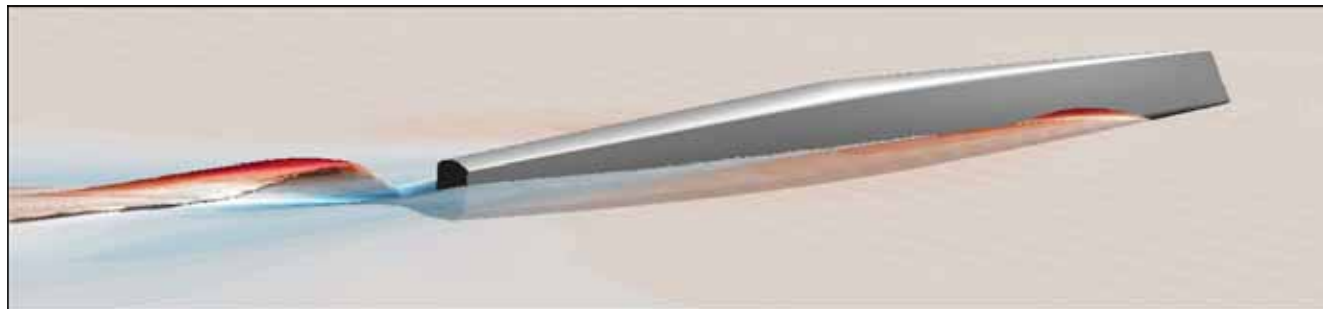


Analytical hull drag model

Analytical models tuned against a database of CFD solutions on the isolated demihull [1]

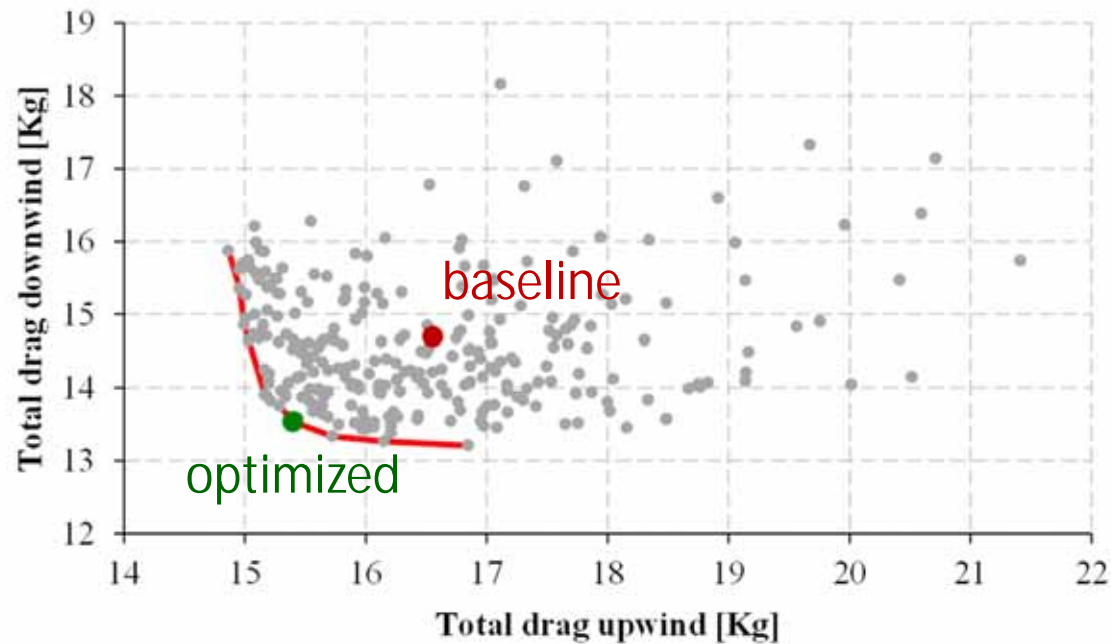


$$D_H = \frac{1}{2} \rho_w V^2 S_{wet} (W_{BO}) [(1 + k) C_f + C_w] [1 + k_\beta V^{\tau\beta} (W_{BO} + w_\beta) \beta^2]$$



[1] Ubaldo Cella, Francesco Salvatore, Raffaele Ponzini, "Coupled Sail and Appendage Design Method for Multihull Based on Numerical Optimisation", PRACE – EU SHAPE Project final report, 5th July 2016, available online at www.prace-ri.eu

Pareto solution

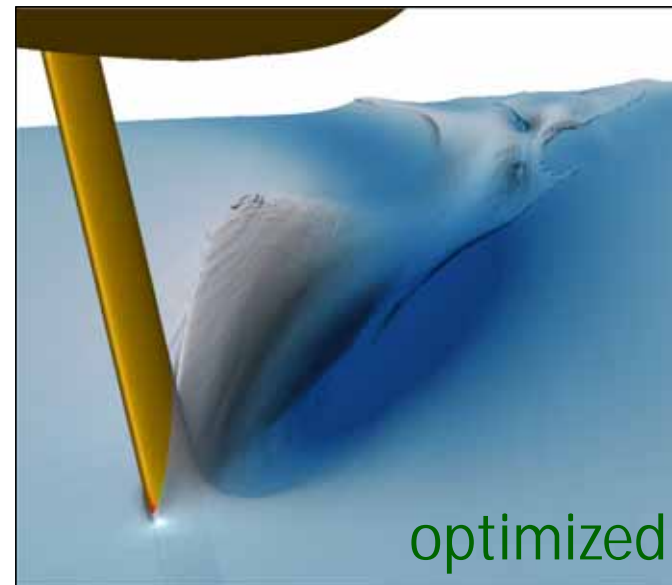
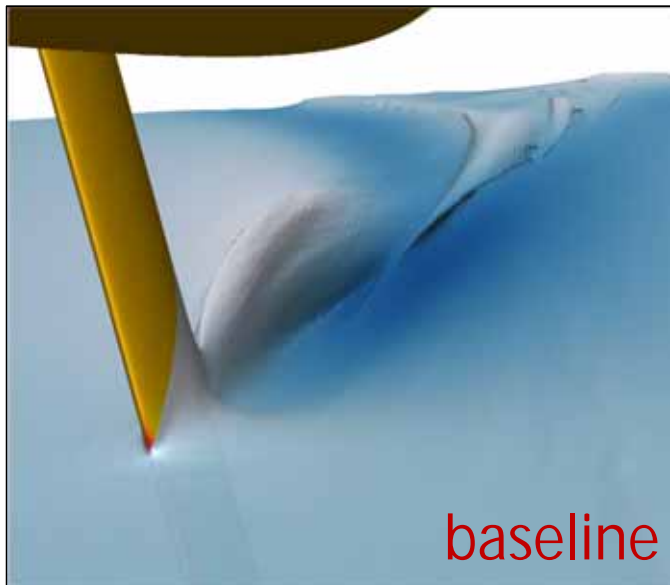


2 objectives
optimization using GAs
Around 400 eval.
Around 40% rejected

Total drag
reduction:
Upwind = - **7** %.
Downwind = - **7.9** %

Post design verification

<i>Mesh</i>	Baseline Kg	Optimized Kg	Drag reduction %
<i>Coarse (1 mill.)</i>	14.7	13.54	7.89
<i>Fine (25 mill.)</i>	13.99	12.92	7.65



Conclusions

- Strongly constrained parameterization problem **successfully faced** by RBF mesh morphing.
- A complex workflow of a test pilot problem was **setup and efficiently integrated** in an optimization environment.
- **Improvement larger than 7%** was obtained starting from a geometry roughly replicating existing designs.

Thank you for your Attention

Ubaldo Cella



(rbf-morph)[™]



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